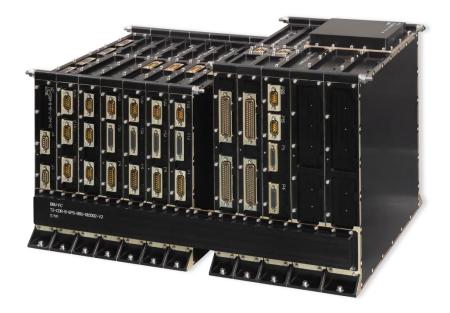




Power Control & Distribution Unit (PCDU)



1. Brief Description

The Power Control & Distribution Unit (PCDU) controls the power flow from the Solar Array and the battery and distributes that power to the whole satellite. It provides efficient power regulation and management to 2 kW of solar power, outputs up to 4 kW of electrical power with protection to bus, payload and all heaters. Its modular and scalable design suits all missions in LEO.

2. Key Features

- In-house Heritage Design
 - 7th year in orbit without anomalies (TRL-9)
 - Upgraded design (TRL-8)
- Compliant to ECSS power interfaces, de-rating & EMC standards
- Highly Reliable and Fault-Tolerant design
 - Single-Point Failure Free
- > 97% Power Conversion Efficiency
- Designed for 4 kW Peak Power in LEO SAR mission
- Optimised for LEO EO or SAR small to medium-sized satellites
 - Adaptable to HEO/MEO missions with parts update
- Configurable 1 A to 10 A LCL Power Distribution Lines
- Modular & Scalable Power System

3. Technical Specifications

General	
System Power	4000 W @ 32 V
Solar Power	2000 W
Payload Power	Up to 3300 W @ 32 V
Primary Bus Voltage	22 ~ 38 V (battery follower)
Solar Array Regulation	Maximum Power Point Tracking (MPPT)
Battery Management	Constant Current (CC) & Constant Voltage
	(CV) Control
Battery Charge Regulators	2 out of 3 redundancy
Power Conversion	> 97%
Efficiency	
Reliability	Single-Point Failure Free architecture
Battery under-voltage	- Disconnect Non-Essential Loads by TC
protection	- Configurable hardware cut-off for
	satellite safe-hold¹

Interfaces	
Latching Current Limiters	56 LCL @ 10 A or less
	48 LCL @ 5 A or less
	4 RLCL (retriggerable) for critical lines
Heater Latching Current	72 heater lines
Limiters	
Thermal Knives	Support 4 mechanisms with ARM & FIRE
Deployment Driver 1	@ 20 V, 8 A
Frangibolt Deployment	Support 4 mechanisms with ARM & FIRE
Driver 2	@ 22 ~ 36 V, 2 A
Solar Array Voltage	42 ~ 100 V
Battery Discharging	Up to 130 A
Current	
TC/TM	Redundant CAN bus to OBC
Separation Switch from	Triple redundancy
Launch Vehicle	

Dimensions & Mass	
Mass	24.5 kg (with current configuration)
Volume	453 x 340 x 279 mm ³
	(with current configuration)

Environments	
Thermal	- 20°C to 60°C operating range
Radiation	TID: 20 kRAD
	SEE: 40 MeV·cm²/mg
Lifetime	5 years
EMI/EMC	MIL-STD-461